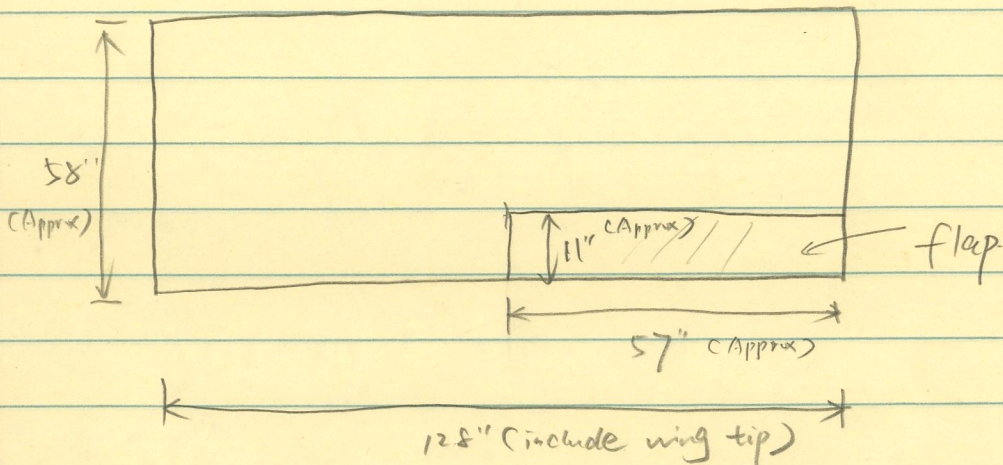


12/14/2011

Flap load calculation.

P. 1.

* Wing and flap dimension



$$\text{Flap force } L_f = C_{L_f} \cdot A_{\text{flap}} \cdot \frac{1}{2} \rho V^2$$

C_{L_f} - lift coeff. of flap

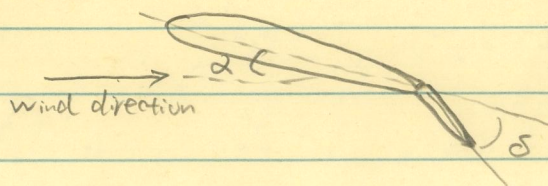
A_{flap} - Area of flap

ρ - density of air

V = speed of airplane

Per study by National Memorial Aeronautical Lab done in 1930.

$$C_{L_f} = 11 C_L - 11 \delta$$



$$E = \frac{C_{\text{flap}}}{C_{\text{wing}}} = \frac{11}{58} = 0.19 \approx 0.2$$

At $E = 0.2$, $n = 0.2$, $n_0 = -2.05$.

Conservatively assuming $C_L = 3$.

flap angle = $40^\circ = 0.7$ rad.

$$\Rightarrow C_{L_f} = n C_L - n_0 \delta = 0.2 \times 3 + 2.05 \times 0.7 = 2.04$$

Air density $\rho = 0.002377$ slug/ft³ at sea level.

Assuming flap extension speed $V_{fe} = 100$ mph = 147 ft/sec

Flap area $A_{flap} = 11'' \times 57'' = 627$ in² = 4.35 ft²

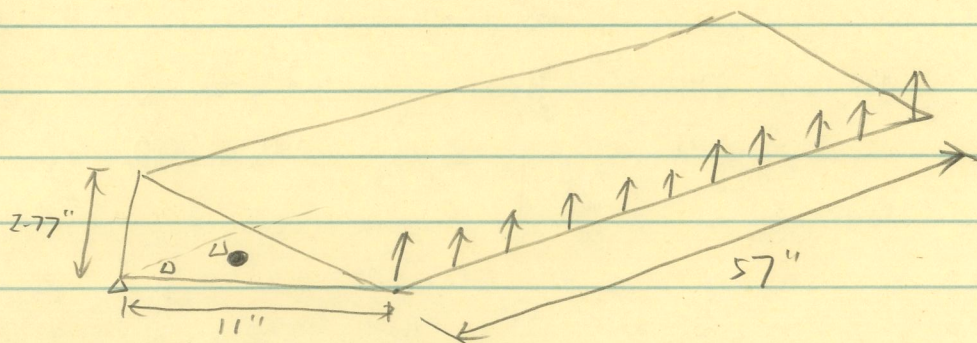
Flap load: $L_f = C_{L_f} \cdot A_{flap} \cdot \frac{1}{2} \rho V^2$

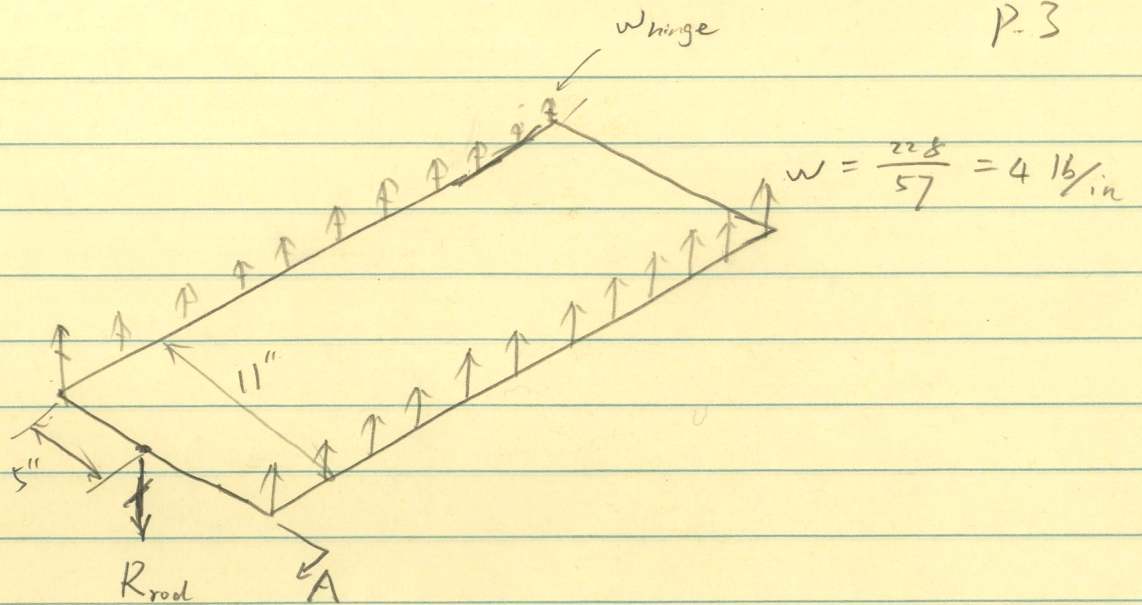
$$= 2.04 \times 4.35 \times 0.002377 \times 147^2 / 2 = \underline{\underline{228 \text{ lbs}}}$$

Now we need to figure out the shear force on the inboard rib rivets. (A)

Assumptions:

- 228 lbs is applied at the trailing edge of the flap (below)



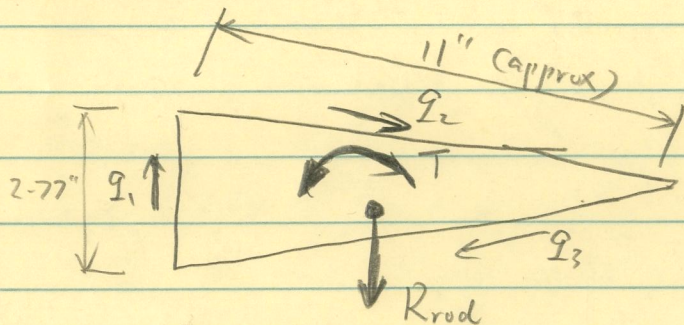


Total moment about hinge line due to flap force

$$M = 228 \times 11 = 2508 \text{ in-lb}$$

$$R_{rod} = 2508 / 5 = 502 \text{ lbs}$$

At section A. (inboard rib section). FBD shown below



$$q_1 = q_{1R} + q_{1T}$$

q_{1R} - shear flow due to R_{rod} . assuming R_{rod} is taken entirely by member 1

q_{1T} - shear flow due to Torque T.

$q_1 = q_2 = q_3 = q_T$, shear flow due to Torque T.

$$q = \frac{T}{2A}$$

Area of cross section A: $A = 11 \times 2.77 / 2 = 15.24 \text{ in}^2$

$$q = 2508 / (2 \times 15.24) = 82.3 \text{ lb/in}$$

Per drawing the rivet spacing at inboard rib is about 1.27"

Shear force at rivet: $P_{\text{shear}} = 82.3 \times 1.27 = 105 \text{ lbs}$

shear allowable of MK-319-B3 rivet.

$$P_{\text{all shear}} = 258 \text{ lbs}$$

$$P_{\text{all tension}} = 292 \text{ lbs.}$$

$$MS = \frac{258}{105} - 1 = 1.45$$

Countersunk all metal

shear Allowable of AN426AD3 for 0.025" sheet = 113 lbs

Tension " " " " 0.026" " = 130 lbs

AN426AD3 shear strength = 207 lbs

tension " =